







25th AIAA International Space planes and Hypersonic Systems and Technologies Conference

Sheraton Grand, Rajaji Nagar Bangalore, India May 28 – June 01, 2023

25th AIAA International Space Planes and Hypersonic Systems and Technologies Conference Program Schedule

	Sunday, 28 May 2023	
PLN1 1730 - 1920 hrs	Welcome Address and Inaugural Panel Discussion: Scientific and Technological Challenges in Hypersonics	Grand Ballroom
M	oderator: Prof. Jagadeesh Gopalan, Chairman, Centre of Excellence in Hypersonics, Indian Institute of Science, India	a.
	Panelists: Dr. Kevin Bowcutt, Principal Senior Technical Fellow and Chief Scientist of Hypersonics, The Boeing Company, USA. , Secretary Department of Defence R&D and Chairman Defence Research and Development Organisation(DRDO), India. Dr. G. Satheesh Reddy, Scientific Adviser to Raksha Mantri, Ministry of Defence, Government of India. h, Secretary Department of Space, Chairman Space Commission & Indian Space Research Organisation(ISRO), Government	
1920 - 1930 hrs	Exhibition Inauguration by Mr. Anil Parab, Whole-time Director, Heavy Engineering and L&T Valves, Larsen and Tubro	Grand Ballroom Foyer
DINNER01 1930 - 2200 hrs	Inaugural Dinner	Grand Ballroom

		Monday,	29 May 2023		
PLN2 - I 0800 - 1000 hrs	18	Country	Reports - I	[3]	Grand Ballroom
Australia: Prof.		esh Gopalan, Chairman, Centre of I In, Associate Professor, School of M			sland, Australia
1000 - 1030 hrs		Monday	Morning Tea	/5/	Grand Ballroom Foyer
PLN2 - II 1030 - 1230 hrs		Country	Reports - II	\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\	Grand Ballroom
		f. Dan Michaels, Associate Professo an: Dr. Kouichiro Tani, Head of Kakı			
LUNCH01 1230 - 1330 hrs		Mond	ay Lunch		Grand Ballroom Foyer
1330 - 1730 hrs		MONDA	AY SESSIONS		
Grand	Ballroom 1	Grand Ballroom 2	Jupiter	Nepti	une
	HYP01 and Vehicles I	HYP02 Propulsion Systems	HYP03 Computational Fluids	HYP0 s I Test and Ev	

		Tuesday, 30) May 2023			
PLN3 0800 - 0930 hrs		Panel Session: Long Duration	Hypersonic System Design		Grand Ballroom	
Moderator: Dr. Anan	М	an, Associate Professor, School of Med Pane Dr. Angus Hendrick, Technical Dire r. Chris Kostyk, Aerospace Engineer, N Ierio Viti, Aerospace and Defense Tec Shri. Satya Swaroop Panda, Direct	el ists: ector, Conventional Prompt Strike IASA Armstrong Flight Research Cent am Manager, Principal Engineer, Ans	er	nsland (invited)	
1000 - 1200 hrs	/5	TUESDAY	SESSIONS I	2		
Grand Ballroc	m 1	Grand Ballroom 2	Jupiter	Neptune		
HYP05 Missions and Vel	nicles II	HYP06 Propulsion Components I		Test	HYP08 and Evaluation I	
LUNCH02 1200 - 1300 hrs	-	Tuesday	Lunch	SO	Grand Ballroom Foye	
1300 - 1500 hrs	14	TUESDAY	SESSIONS II	3		
Grand Ballroc	m 1	Grand Ballroom 2	Jupiter	Neptu	ıne	
HYP09 Hypersonic Funda	mentals I	HYP10 Propulsion Components II	HYP11 Computational Fluids II	HYP [*] Test and Evo		
EXCURSION1 1500 - 1800 hrs			personic Labs Visit, Tea ded to/from the Conference venue			
CULTURAL Evening 1845 - 2000 hrs		Felicitations & C Performance by Laya Lavanya Karno Bharatnatyam by Bha	ataka Classical Percussion Ensemble		Granc Ballroom	
DINNER02 2000 - 2230 hrs		Hypersonic	s Banquet		Grand Ballroom Foye	

	Wednesday,	31 May 2023	
PLN4 0800 - 0930 hrs	Panel Session: Latest Developm	ents in Hypersonics from India	Granc Ballroom
Moderator: Prof. Srisho	Pane Pane Dr. S. Unnikrishnan N Shri G A S Murthy Dr. V Ashok, Deputy D Dr. T K Ganesh Anavardham, I Dr. Ajith Chaudary, Proje Mr. Laxmesh B H, VP, Head of Missiles	elists: Nair, Director VSSC v, Director DRDL Dir <mark>ector</mark> – Aero, VSSC Project Director HSTDV, DRDL ct Director, HCM, DRDL	
930 - 1000 hrs	Wednesday I	Morning Tea	Grand Ballroom Foye
000 - 1230 hrs	WEDNESDAY	SESSIONS I	3
Grand Ballroom 1	Grand Ballroom 2	Jupiter	Neptune
HYP13 Hypersonic Fundamentals II	HYP14 Propulsion Components III	HYP15 Computational Fluids III	HYP16 Thermal Management Systems
UNCH03 230 - 1330 hrs	Wednesdo	ay Lunch	Grand Ballroom Foye
330 - 1700 hrs	WEDNESDAY	SESSIONS II	
Grand Ballroom 1	Grand Ballroom 2	Jupiter	Neptune
HYP17 Hypersonic Fundamentals III	HYP18 Other Advanced Propulsion Topics	/6	HYP20 Materials and Structures
700 - 1730 hrs	VALEDICTOR	Y FUNCTION	Grand Ballroom
		y High Tea	Grand

	Thursday, 1 June 2023
EXCURSION2	Conference Excursion to Mysore
0800 - 1900 hrs	Reporting time at 0730 hrs

HYP01 1330 – 1730 hrs			Mis	sions and Vehic	les I	Session	s' Details	Grand Ballroom 1
1330 hrs AIAA-2023-3000 Aerodynamic analysis of a flap-based deployable re- entry system in different flight conditions E. Gaglio, Scuola Superiore Meridionale, Napoli, Italy; A. Cecere, R. Guida, S. Mungiguerra, R. Savino, Universita degli Studi di Napoli, Italy; N. Chaqueo Jara, Universidad Tecnica Federico Santa Maria, Valparaiso, Chile; et al.	1350 hrs AIAA-2023-3001 Safe and Adaptive Trajectory Reshaping of Constrained Re- entry Flight: Recovery Ensemble Control A. Selim, Roketsan Roket Sanayii ve Ticaret AS, Ankara, Turkey; I. Ozkol, Istanbul Teknik Universitesi Ucak ve Uzay Bilimleri Fakultesi, Istanbul, Turkey	1410 hrs AIAA-2023-3002 Potential Aerodynamics Benefits of the Camber- Morphing Wing D. Charles F, P. Kumar, S. Muzzammil, A. Prabhu, K. Shanker, RV College of Engineering, Bangalore, India	1430 hrs AIAA-2023-3003 Experimental Studies to Design of Robust and Optimal Scramjet Engine Intake Configuration N. Joseph, S. S, F. K J, S. Chandran R, M. B, Vikram Sarabhai Space Centre, Thiruvananthapuram India	STAJe 1 Payload J. Kunze, A. Paull, The University of Queensland, Saint Lucia, Australia	d preliminary design: Aerodynamic	Affordability and Flexibility in Access to Space		
HYP02 1330 – 1730 hrs		19	P	ropulsion System	าร	· / 0		Grand Ballroom 2
1330 hrs AIAA-2023-3007 Preliminary Study on SABRE Engine A. Hale, MET Institute of Engineering, Nashik, India	1350 hrs AIAA-2023-3008 Flame- Stabilization of Liquid Hydrocarbon Fuel and Gaseous Surrogates in a Scramjet Combustor J. van der Lee, D. Michaels, J. Lefkowitz, Technion Israel Institute of Technology, Haifa, Israel	1410 hrs AIAA-2023-3009 JAXA RD1 Flight Experiment on Supersonic Combustion: Part 1. Overview K. Tani, M. Takegoshi, K. Takasaki, S. Tokudome, Uchu Koku Kenkyu Kaihatsu Kiko, Kakuda, Japan	AIAA-2023-3010 Influence of Inflection Mach Number and Base Nozzle Length on the Shock Vector Control of a Planar Nozzle with Double Divergence A. Das, T. Mankodi, U. Saha, Indian Institute of	AIAA-2023-3011 JAXA RD1 Flight Experiment on Supersonic Combustion: Part 2 Combustion Test Result M. Takahashi, K. Kobayashi, M. Takegoshi, T. Saito,	1510 hrs AIAA-2023-3012 Numerical Analysis of an integrated Scramjet vehicle at Hypersonic Speed A. Singh, Vikram Sarabhai Space Centre, Thiruvananthapuram, India	Investigation of Flame Flashback Phenomenon in a Strut-Cavity Based Scramjet Combustor S. Pranaykumar, D.	1550 hrs AIAA-2023-3014 Nitrogen Oxide Formation Pathways in Gaseous Detonations A. Dahake, R. Singh, A. Singh, Indian Institute of Technology Kanpur, Kanpur, India	1610 hrs AIAA-2023-3015 Intake starting characteristics of a scramjet combustor M. Mahato, S. Chimakurthy, C. Jalla, R. Pahade, S. Chattopadhayay, G. T K, DRDL, Hyderabad, India; et al.

HYP03 1330 – 1730 hrs			Cor	nputational Flui	ds I							Jupiter
Distance and Development of a Supersonic Turbulent Boundary Layer: CFD Analysis of a Diamond- shaped 2D Fin C. Morency, K. Jansen, University of Colorado Boulder, Boulder,	Assessment of Modified \(\gamma\)- model for Hypersonic Boundary Layer Transition Prediction Considering Leading Edge Bluntness D. R C, A. K, R. G, Indian Institute of	AIAA-2023-3018 Framework to Characterize Aeroacoustic Loads Beneath High-Speed Boundary Layers S. Kaluva, R. Kumar, N. Vadlamani, Indian Institute of Technology Madras, Chennai, India	nvestigation of flow Phenomena in the Optimization of Eupersonic Intake Design with a RANS Follows. Singh Sandhu, M. Shardwaj, A. Shard	1 450 hrs AIAA-2023-3020 A Shock Fitting Technique For Hypersonic Flows Using Hexahedral Meshes K. Baliga, R. Krishnamurthy, Program Development Company LLC, ndia, Bangalore, ndia; M. Spel, R. Tech Research and Technology, Verniolle, France; S. James, V. Soudharshanam, Program Development Company LLC, ndia, Bangalore, ndia, Bangalore, ndia, Bangalore, ndia	Inverse of- Charander Algo Unster Flows Shiffi Therr Equil A. Jan The Universe Universe Universe Universe	A-2023-3021 rse Method- racteristics orithms for eady Gas s with	Aerody Shape Optimi Hypers Missile B. Prasc Padma goyal, J Sivasub M. S. Ro Universi Applied	o23-3022 ynamic zation of a onic Geometry d, M. nabha, p. ramanian, maiah	Aero-The Environn Apollo E Test, FIRE Adaptive Refinem	23-3023 ration rynamics on of ermal ment for fra Flight E-II using e Grid ment and mperature tava, I. b. tra, P. man, S. S, c, Pune,		
HYP04 1330 – 1730 hrs		8/		Test and Evalu	ation			13		•		Neptune
1330 hrs AIAA-2023-3024 Design and Viscous Correction of an Axisymmetric Contoured Nozzle for Perfect Gas to the T2 Hypersonic Shock Tunnel R. Vilela, A. Fraile Junior, L. Ribeiro, R. Santos, P. MATOS, I. Rego, Instituto de Estudos Avancados, Sao Jose dos Campos, Brazil; et al.	Ethylene Fuel Injection System in the T4 Hypersonic Impulse Facility. M. Trudgian, H. Russell, T. Vanyai, A. Veeraragavan, The University of Queensland	1410 hrs AIAA-2023-3026 Measuring the Free-Stream Turbulence Spectrum in the Shock Tunnel HELM using the Focused Laser Differential Interferometry (FLDI) T. Sander, L. Jakobs, C. Mundt, Universitat der Bundeswehr Munchen Fakultat fur Luft- und Raumfahrttechnik, Neubiberg, Germany	Mach 10 Flight Conditions M. Minucci, L. Galembeck, D. Carinhana Jr., I.	te implementation extended test times J. Higgs, F. Arafin Pellegrini, A. Aguilera, K. Rou R. Rahman, Univo of Central Florid Orlando, FL; et a	on for n, J. viere, versity a,	Hypersonic N Tunnel Exper H. Takahashi, . Hirotani, H. Tag Japan Aerosp	c on of craft ields in Wind criment J. Oki, T. guchi, ace gency,	1530 hrs AIAA-2023 Numerica Analysis o Oxygen Ri Gas Dispo during Gro Test of Sen cryogenic Subsystem A. Singhal, I Agarwal, T. Tharakan, S S, Liquid Pro Systems Ce Valiamala,	f fch Hot sal bund ni- Engine n D. . Kumar opulsion ntre,	1550 hrs AIAA-2023- Measurem of Free stre Density Fluctuation the T4 Stalk Tube R. Ananthapaa aban, A. Veeraragav McIntyre, V. Wheatley, D The Universit Queensland Lucia, Austra	ents eam as in cer dman an, T. b. Mee, y of l, Saint	

HYP05 1000 – 1200 hrs			Miss	ions and Vehicle	es II		Grand Ballroom 1
characteristics in Scramjet using Back Pressure fluctuations at	Effects of Aerothermal Shape Distortion on Hypersonic Vehicle Performance in Cruise L. Pollock, J. Moran, A. Neely,	AIAA-2023-3034 Conceptual Study into Wave Ingesting Propulsion for Hypersonic Flight I. Jahn, University of Southern Queensland, Toowoomba, Australia	Aerodynamic Characterization of Hypersonic Launch Vehicle laden with exposed Scramjet based	1120 hrs AIAA-2023-3036 SHAR for a TSTO launcher N. Relangi, L. Peri, A. Ingenito, SCUOLA DI INGEGNERIA AEROSPAZIALE, SAPIENZA, ROMA, Italy	E PLAZZ	ANT OF THE	
HYP06 1000 – 1200 hrs		55	Prop	ulsion Compone	nts I	I FI	Grand Ballroom 2
Influence of Active Cooling on Engine Performance by Mitigating Aero- thermo-elastic Deformation of a Hypersonic Inlet	1020 hrs AIAA-2023-3038 Experimental and Numerical Investigation of Leading-Edge Bluntness on Shock Boundary layer Interaction Inside a Scramjet Inlet at Mach 6 T. Vamsi krishna, J. Sreekumar, Indian Institute of Technology Kanpur, Kanpur, India; D. SLN, Vikram Sarabhai Space Centre, Thiruvananthapuram India; M. Sugarno, Indian Institute of Technology Kanpur, Kanpur, India	K. Hyunwoo, J. Kang, H. Sung, Korea Aerospace University, Goyang, South Korea	losses nature in a Scramjet Combustor S. Palateerdham, L Peri, A. Ingenito, SCUOLA DI	Supersonic Crossflow using a Homogeneous Mixture Model with AMR Y. Donggyu, H. Sung, Korea Aerospace University, Goyang, South Korea	1140 hrs AIAA-2023-3042 Analysis of unsteady pressure measurements in the unstart of a Mach 6.58 hypersonic intake N. V, S. Rao, Indian Institute of Science, Bangalore, India	SONIC STSTER	

HYP08 1000 – 1200 hrs			Test and Evaluation II	Neptune
An Overview of Recent Advances in Flow Characterization Utilizing Laser-Based Diagnostics in the ONR-UTA Arc-Heated	A. Mahanti, S. K., D. Mishra, R. Sadanandan, Indian Institute of Space Science and	Optical Emission Spectroscopy and Thomson Scattering Diagnosis on Laser-Induced Ablated Plume from HfB ₂ -SiC Material J. George, MetroLaser Inc, Laguna Hills, CA; Y. Wu, C. Limbach, Texas A&M University,	1100 hrs AIAA-2023-3046 One-dimensional Interferometric Scattering Velocimetry for High-speed Flows Y. Krishna, A. Sekar, Indian Institute of Space Science and Technology, Thiruvananthapuram, India; G. Magnotti, King Abdullah University of Science and Technology, Thuwal, Saudi Arabia; A. Vaidyanathan, Indian Institute of Space Science and Technology, Thiruvananthapuram, India	
HYP09 1300 – 1500 hrs		*	Hypersonic Fundamentals I	Grand Ballroom 1
over a 34° compression corner using OpenFOAM® A. Makhija, K. Bodi, Indian Institute of Technology	simulation of heat transfer for a diamond airfoil in hypersonic flow A. Makhija, K. Bodi, T. Chandra Sekar, Indian Institute of Technology Bombay, Mumbai, India	1340 hrs AIAA-2023-3049 A Method for Solving Oblique- Shock/Expansion- Fan Interaction N. Doshi, Indian Institute of Technology Kharagpur, Kharagpur, India; M. Devaraj, G. Jagadeesh, Indian Institute of Science, Bangalore, India	SENO TONHO LE CHINO OCIVE	

HYP10 1300 – 1500 hrs		Propulsion Components II	Grand Ballroom 2
1300 hrs AIAA-2023-3050 Supersonic combustion supported by flameholders L. Vialta, Instituto Tecnologico de Aeronautica, Sao Jose dos Campos, Brazil; P. Matos, L. Barreta, D. Carinhana, F. Eugênio Ribeiro, Instituto de Estudos Avancados, Sao Jose dos Campos, Brazil	1320 hrs AIAA-2023-3051 Effect of Leading Edge Bluntness on Aerothermal Characteristics of Osculating Cone Waveriders A. Ghosh, A. Rao, S. Rao, Indian Institute of Science, Bangalore, India 1340 hrs AIAA-2023-3052 Design Methodology for Selecting a Single Expansion Ramp Nozzle Geometry that can Complete of Mission between Machs 3 and 7 D. Cerantola, J. Gagnon, D. Handford, P. Dass, Space Engine Systems, Edmonton, Canada	Induced Mixing Enhancement for Transverse Injection in Scramjet Engines C. Fujio, H. Ogawa, Kyushu Daigaku, Fukuoka, Japan an Internal Cross- Flow Effervescent Ramp Injector S. Dharan, A. Vaidyanathan, Indian Institute of Space Science and Technology Department of Aerospace	
HYP11 1300 – 1500 hrs	1.2	Computational Fluids II	Jupiter
1300 hrs AIAA-2023-3055 Shape Optimization of Hypersonic Inlet using a new Clustering-based Hybrid Optimizer. K. Bahuguna, Indian Institute of Science, Bangalore, India; R. Kolluru, Cambridge Centre for Advanced Research and Education in Singapore, Singapore; S. Raghurama Rao, Indian Institute of Science, Bangalore, India	family schemes on a hybrid	Two dimensional laminar flow computations on unstructured meshes using defect corrected geometries	

HYP12 1300 – 1500 hrs			Te	st and Evaluatio	n III			Neptune
1300 hrs AIAA-2023-3060 Boundary Layer Transition Studies on a Winged Body Reentry Vehicle through Heat Flux Measurements in Shock Tunnel R. Ravichandran, A. Ahmed, S. R, A. Kumar, D. SLN, M. B, Vikram Sarabhai Space Centre, Thiruvananthapuram, India; et al.	intake flowfield and effect on combustion. G. Pelletier, G. Vilmart, a. Vincent- Randonnier, C. Brossard, Office National d'Etudes	Numerical analysis of FBG response to blast wave inside shock tube G. Hegde, G. Hegde, G. Jagadeesh, S. Asokan, Indian Institute of Science, Bangalore, India	1400 hrs AIAA-2023-3063 Investigation of th unsteadiness during starting transient operation of a zero secondary flow ejector. M. Raju, Vikram Sarabhai Space Centre, Thiruvananthapuran India	Analysis of Shock Propagation from Shock Tuk for Blast Testing A. Jraisheh, J. Chutia, A. Patha V. Kulkarni, Indian Institute of	р е К,	THO HAP		
HYP13 0930 - 1230 hrs		£ 2 ₽	Hyper	sonic Fundame	ntals II	믺		Grand Ballroom 1
AIAA-2023-3065 Study of Shock/Boundary-Layer Interactions Generated by a Sharp Fin Mounted above a Body of Revolution D. Otten, Lockheed Martin Missiles and	Study of Spanwise Dual Jet Injection in a Supersonic	the Hypersonic Intake in Varied Free-Stream Conditions A. Ray, A. De, Indian Institute of	Effects of enthalpy on convective Heating of a large angle blunt cone in Martian atmosphere S. Bhaduri, A. De, M. Sugarno, Indian Institute of Technology Kanpur, Kanpur, India; D. SLN, Vikram Sarabhai Space Centre, Thiruvananthapuram,	1050 hrs AIAA-2023-3069 A parametric study on a cone-cavity model in a laminar hypersonic flow S. Nanda, Technion Israel Institute of Technology, Haifa, Israel; K. SK, Indian Institute of Technology Madras, Chennai, India; J. Cohen, Technion Israel Institute of Technology, Haifa, Israel	1110 hrs AIAA-2023-3070 Phenomenological Introduction of Standoff Distance of Sonic Ring Impelling Entropy Waves and Aerodynamic Heating of Hypersonic Vehicles V. Sanal Kumar, Vikram Sarabhai Space Centre, Thiruvananthapuram, India; D. Vinay, R. Sharma, S. Sharma, Amity University, Noida, India; V. Rajendran, The University of New South Wales Canberra, Canberra, Australia; R. Sarswat, Amity University, Noida, India; et al.	speeds M. Talluru, S. Gai, University of New South	1150 hrs AIAA-2023-3072 Response of a cantilevered plate to shock impingement in hypersonic flows K. Ahuja, S. Rao, Indian Institute of Science, Bangalore, India	

HYP14 0930 - 1230 hrs	Propulsion Components III						Grand Ballroom 2
0930 hrs AIAA-2023-3073 Atomization and Mixing of Liquid and Aerated jet injected behind Curved Pylon in Supersonic Cross- flow A. Sekar, Indian Institute of Space Science and Technology Department of Aerospace Engineering, Thiruvananthapuram, India; Y. Krishna, Specrule Scientific Pvt Ltd, Thiruvananthapuram, India; A. Vaidyanathan, Indiai Institute of Space Science and Technology Department of Aerospace Engineering, Thiruvananthapuram, Indian Institute of Space Science and Technology Department of Aerospace Engineering, Thiruvananthapuram, India			1030 hrs AIAA-2023-3076 Experiments on mass injection- induced unstarting flows M. MS, N. Thakor, M. Devaraj, S. S, Indian Institute of Science, Bangalore, India	1050 hrs AIAA-2023-3077 Influence of cavitation on the fundamental flame studies of surrogate fuels: A potential realization towards supersonic combustion. b. Rajaghatta Sundararam, MS Ramaiah University of Applied Sciences, Bangalore, India; P. Panda, Indian Institute of Science, Bangalore, India; D. S Arulumallige, V. M Shelar, MS Ramaiah University of Applied Sciences, Bangalore, India; D. S Arulumallige, V. M Shelar, MS Ramaiah University of Applied Sciences, Bangalore, India	Impulse Facility A. Ahmed, N. Ameen K, S. R, B. Mathew K, D. SLN, M. B, Vikram Sarabhai Space Centre, Thiruvananthapuram, India; et al.	PE	

HYP15 0930 - 1230 hrs	Computational Fluids III							
0930 hrs AIAA-2023-3079 On Residual Estimate of Local Truncation Error	0950 hrs AIAA-2023-3080 Hypersonic simulation of a free-flying ring interfering with a two-dimensional curved shock wave M. Chemak, ISAE- ENSMA, Futuroscope Chasseneuil, France; M. Spel, R.Tech, Verniolle, France; S. James, Program Development Company, Kodihalli, India; P. Eiseman, Program Development Company, White Plains, NY		University of Kentucky, Lexington, KY	1050 hrs AIAA-2023-3083 Hypersonic Aerodynamic Characterization of a Winged Body Re-entry Configuration Using Chemical Non-Equilibrium CFD Simulations A. Nagpal, V. G, Vikram Sarabhai Space Centre, Thiruvananthapuram, India	for the HIFIRE-1 Flight Research Vehicle V. V, J. Sivasubramanian, M. S. Ramaiah University of Applied Sciences, Bangalore, India	1130 hrs AIAA-2023-3085 Prediction of drag over a bluff body by coupling of artificial neural network with CFD based upstream energy deposition method A. Pathak, P. Khare, J. Chutia, A. Jraisheh, V. Kulkarni, Indian Institute of Technology Guwahati, Guwahati, India	1150 hrs AIAA-2023-3086 Linear Stability Analysis of a Hypersonic Boundary Layer on a Flat Plate J. Sivasubramanian, M. S. Ramaiah University of Applied Sciences, Bangalore, India	
HYP16 0930 - 1230 hrs		181	Therm	nal Management S	ystems			Neptun

HYP16 0930 - 1230 hrs	Thermal Management Systems							Neptune
AIAA-2023-3087 Film Cooling Effectiveness on Radiative Heat Transfer in Hypersonic Flows J. Sreekumar, T. Vamsi krishna, M. Sugarno, Indian Institute of Technology	of a Purged Cryogenic CFRP Tank and TPS Test Demonstrator for RLV T. Reimer, C. Rauh,	Hypersonic Flight Experiment Thermal Protection System T. Reimer, G. Di Martino, I. Petkov, L. Dauth, L. Baier, A. Gülhan, Deutsches Zentrum fur Luft- und Raumfahrt eV, Stuttaart. Germany	Transfer in a Scramjet Forebody under Varying Hypersonic Flow G. Nagarajan Kirupakaran, Indian Institute of Science, Bangalore, India; G. K V, BrahMos	1050 hrs AIAA-2023-3091 Manufacturing and Testing of a Purged Thermal Insulation Concept for Reusable CFRP Cryotanks C. Rauh, T. Reimer, German Aerospace Center (DLR), Stuttgart, Germany	Panda, Indian			

HYP17 1330 - 1700 hrs	Hypersonic Fundamentals III							Grand Ballroom 1
Oscillating shock wave boundary layer interactions on a cantilever plate M. Talluru, L. McQuellin, A. Neely, University of New South Wales Canberra at ADFA.	Experimental Investigation of the Shock Wave/Boundary Layer Interaction at a Compression Corner Model A. Lara, T. Rolim, L. Barreta, P. MATOS,	Leading Edge Flat Plate in a Hypersonic Shock Tunnel A. Bajpai, G.	Signals to Reconstruct the Shape of the Plasma Sheet Surrounding the Re-Entry Vehicles G. Palmerini, P.	1450 hrs AIAA-2023-3098 Noise Prediction for Mach 8 Waverider vehicle during Take-Off and Landing G. Piccirillo, N. Viola, R. Fusaro, Politecnico di Torino, Torino, Italy	Aircraft Corporation of	1530 hrs AIAA-2023-3100 Effect of Isolator length in Nozzle - Isolator interaction in co- flow set up recurrent in hypersonic propulsion systems A. Himakar, R. K, S. Rao, Indian Institute of Science, Bangalore, India		

HYP18 1330 - 1700 hrs	Other Advanced Propulsion Topics hrs							
1330 hrs AIAA-2023-3101 Study of Aerospike Nozzle and its Thrust Vectoring Characteristics V. Nagaral, R. R, G. Narayana, MS Ramaiah University of Applied Sciences, Bangalore, India	1350 hrs AIAA-2023-3102 Numerical Studies of Wall Temperature effects on Shock-Wave/ Turbulent Boundary Layer Interaction over Annular Conical Plug Nozzle H. Gahnolia, R. Pal, A. Roy, Indian Institute of Technology Kharagpur, Kharagpur, India	1410 hrs AIAA-2023-3103 Enhancing the Detonability of H ₂ -O ₂ /Air Mixtures by Using O ₃ and H ₂ O ₂ for Application in Detonation Engines R. Singh, A. Dahake, A. Singh, Indian Institute of Technology Kanpur, Kanpur, India	1430 hrs AIAA-2023-3104 A Neural Network Based Design of a Planar Double Divergent Nozzle K. Shrivastava, A. Das, U. Saha, Indian Institute of Technology Guwahati, Guwahati, India	1450 hrs AIAA-2023-3105 In Silico Studies on Truncated Aerospike Nozzle with Optimum Cowl Length for Single Stage to Orbit Vehicles S. Sundaria, A. Bhagat, Amity University, Noida, India; V. Sanal Kumar, Vikram Sarabhai Space Centre, Thiruvananthapuram, India	1510 hrs AIAA-2023-3106 Addition of Hydrogen to Enhance Ignition Characteristics and Performance of a Rocket Propellant for Application in Detonation Engines R. Singh, A. Dahake, A. Singh, Indian Institute of Technology Kanpur, Kanpur, India	1530 hrs AIAA-2023-3107 Effect of Ozone and Inert Diluents on RP1/RP2 – O2 Gaseous Detonations for Applications in Detonation- based Rocket Engines S. Naraboina, R. Singh, A. Dahake, A. Singh, Indian Institute of Technology Kanpur, Kanpur, India	1550 hrs AIAA-2023-3108 Sub-scale experimental test of Hybrid Rocket engine using LOX and 3D printed fuel S. Patial, A. Rai, D. Sojitra, G. Ayyappan, A. Vaidyanathan, Indian Institute of Space Science and Technology Department of Aerospace Engineering, Thiruvananthapuram, India; D. Bhalla, Vikram Sarabhai Space Centre, Thiruvananthapuram, India	

HYP20 1330 - 1700 hrs	Materials and Structures							Neptune
A Study on Multi- wall Carbon Nanotubes for Aviation Vehicle Structure Suitability in Flight Regimes S. Tarale, S. B, BMS College of	1350 hrs AIAA-2023-3110 Study On Fundamental Lamb Wave Mode Distortions Generated Using Piezoelectric Transducer S. Neharkar, B. C.R, Indian Institute of Space Science and Technology, Thiruvananthapuram, India	1410 hrs AIAA-2023-3111 Linear Lamb Wave Based Technique for Damage Detection in stiffened composite plate using Machine learning S. Kumar, V. MISHRA, M. SUNNY, Indian Institute of Technology Kharagpur, Kharagpur, India	1430 hrs AIAA-2023-3112 Effect of Crack on Strength and Stiffness of Composite Structures using VAM K. Sahu, Indian Institute of Science, Bangalore, India	1450 hrs AIAA-2023-3113 Strength Estimation of Fiber Metal Laminated Plates (GLARE) with Crack by VAM R. Kaushik, Indian Institute of Science, Bangalore, India	E PLANS	SAMOHY		

Turbulent Shear Flow Physics and Engineering Laboratory (TSFPEL)

The Turbulent Shear Flow Physics and Engineering Laboratory at IISc in engaged in experimental research over a wide range of Mach numbers, starting from a low subsonic value and going up to Mach 10. Four different wind tunnels allow us to span this Mach number range. The largest of the four facilities is the Roddam Narasimha Hypersonic Wind Tunnel (RNHWT), a 0.5 m diameter enclosed free-jet facility, which can be operated between Mach numbers 6 and 10. The RNHWT facility will be showcased during the tour.

IISc Hypersonic Labs' Visit





May 30th 2023 1500 hrs to 1800 hrs

Combustion Research and Advanced Diagnostics Laboratory (CRADL)

Combustion Research and Advanced Diagnostics Laboratory (CRADL) deals with supersonic combustion experiments. This facility can handle mass flow rate of fuel from 0.6 to 1.5 kg/s. Static temperature and pressure varies from 800 K to 1050 K and 0.5 bar to 1 bar respectively. Total temperature pressure are in the range of 1600 K to 1800 K and 5.3 bar to 10 bar. In these conditions, a run time of 13 seconds can be achieved. To operate at higher Mach numbers, liquid fuel heater can be incorporated which uses Methane or hydrogen as fuel.

Laboratory For Hypersonic and Shockwave Research (LHSR)

Laboratory for Hypersonic and Shockwave Research (LHSR) is a place of interdisciplinary research varying from hypersonic aerodynamics to various applications of shock waves. Shock tubes form an important asset of the lab for carrying out experiments and shock tunnels that can operate in the range of Mach 6 to Mach 14 with specific enthalpy reaching as high as 5 MJ/kg help in achieving desired flow conditions. The tunnel test sections can accommodate models as large as 100 mm in diameter and 1 m in length. The Free Piston Driven Shock Tube(FPST) that uses free piston to compress gas in the driver tube is a state of the art facility that helps achieve higher enthalpies. Impact of shock waves on chemical kinetics of fuels, materials response, medical applications, industrial technology development etc., have been some of the interdisciplinary areas of the research of this lab. In this visit various facilities of the lab shall be shown.





Cultural Evening

May 30th 2023 1845 hrs to 2000 hrs

Get enchanted into the range of traditional, fusion and folk melodies played by "Laya Lavanya Karnataka Classical Percussion ensemble" consisting of Flute, Violin, Mrudangam, Tabla, Phakwaj, Rhythm pad, Morching, Kanjari and Konnakhol.

Get mesmerized at the three compositions directed by Shri. Sheela Chandrasekhar in Bharatanatyam, one of the traditional dance forms of India by "Bhanumati's Bharatanjali".

The troupe would perform three compositions 1) Poorvarangavidhi, 2) Subrahmanya Namaste, and 3) Sandisha me Shanthim enacting both the beauty in traditions and the vision of modern India by Mahatma Gandhi, Father of Nation, India









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